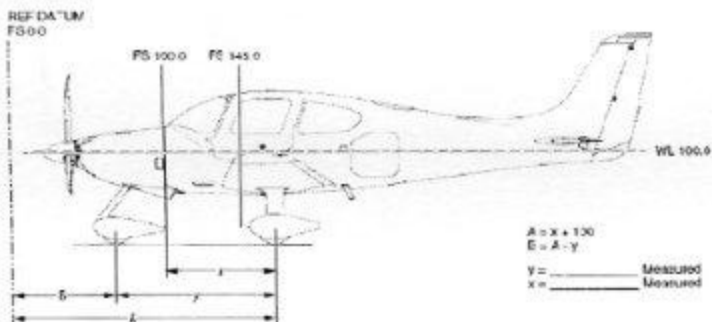


# Airplane weighting sheet



AUTHORIZED  
SERVICE CENTER

Registration	N342CP	S/N	3042
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Series 0032 thru 2933, 2935 thru 2419, 2421 thru 2407

SF02, MM01, 134 B

Weighing Point	Scale Reading (kg)	- Tare	= Net Weight (lb)	X Arm (inch)	= Moment
Left Main			0,0	A =	0,000
Right Main			0,0	A =	0,000
Nose Gear			0,0	B =	0,000
Total as Weighted	0	0	2 373,80	CG = 138,9	329 720,820
CG = Total Moment / Total Weight					
Space below provided for addition or subtraction to as weighted conditions					
Removed aluminium propeller			-79,80	Arm = 48	-3 830,400
Installed composite propeller			67,20	Arm = 48	3 225,600
				Arm =	0,000
				Arm =	0,000
Empty Weight			2 361,20	CG = 139,39	329 116,020
<b>Basic empty weight</b>			<b>2 361,20</b>	<b>CG = 139,39</b>	<b>329 116,020</b>
C.G. % MAC = 100 × (C.G. Inches - LEMAC) ÷ MAC = 100 × (C.G. Inches - 133.1) ÷ 47.7					13,18%

**Weighted on:** 17/09/2008



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